

It reaches the insertion and determines its tense state. There are many technical problems here, for example, insertion fluctuations. In any case, the initial urgent task is to study the features and critical values of stress fields in the perturbed region [24].

In mathematical modeling of the problem of underground blasting, the soil is considered as an elastic medium occupying a quarter of the plane, and the insert as an elastic structure. The shock load is represented as a harmonic distributed load that acts along the normal to the vertical surface of the quarter plane on a certain section. In the Cartesian coordinate system x_i ($i = 1, 2$) the rectangular insert (body 1) occupies the area $D_1(0 \leq x_1 \leq L_2 \cap |x_2| \geq l)$ and the quarter plane (body 2) is the area $D_2(L_1 \leq x_1 \leq \infty \cap -\infty \leq x_2 \leq L_3)$ (Fig. 1), and the quarter plane (body 2) is the $D_2(L_1 \leq L_2; l \geq L_3)$.

The boundary value problem assumes that the initial:

$$v_\alpha^{(k)} = \sigma_{\alpha j}^{(k)} = 0, (\alpha, j = 1, 2; k = 1, 2), \quad (1)$$

boundary conditions:

$$\sigma_{1j}^{(2)} = 0, (j = 1, 2) \text{ by } x_1 = L_1, l \leq |x_2| < \infty \quad (2)$$

$$\sigma_{j2}^{(1)} = 0, (j = 1, 2) \text{ by } 0 \leq x_1 \leq L_1, |x_2| = l \quad (3)$$

and contact conditions:

$$v_\alpha^{(1)} = v_\alpha^{(2)}, \sigma_{1j}^{(1)} = \sigma_{1j}^{(2)}, (\alpha, j = 1, 2) \text{ by } x_1 = L_2, |x_2| \leq l \quad (4)$$

$$v_\alpha^{(1)} = v_\alpha^{(2)}, \sigma_{j2}^{(1)} = \sigma_{j2}^{(2)}, (\alpha, j = 1, 2) \text{ by } L_1 \leq x_1 \leq L_2, |x_2| = l \quad (5)$$

At the boundary EF ($x_1 = 0, |x_2| \leq 1$), the stress-free surface conditions are assumed (Fig. 1).

$$\sigma_{11}^1 = 0, \sigma_{12}^1 = 0 \quad (6)$$

It is assumed that the NB boundary NB ($0 \leq x_1 \leq \infty, x_2 = L_3$) is also free from stresses (3) and only on the local section ($N_1 \leq x_1 \leq N_2, x_2 = L_3$) of the NB boundary of the quarter plane, the normal component σ_{22}^2 acts at certain time intervals, changing according to the la.

$$\sigma_{22}^2(t) = \begin{cases} 0, t \leq 0 \text{ or } S_i \leq t \leq t_i^* \\ A_0 \sin\left(\frac{\omega(t-S_i)}{B_0}\right), t_{i-1}^* \leq t \leq S_i \end{cases} \quad (7)$$

$$\sigma_{12}^2(t) = 0,$$

where $i = 1, 2, \dots, n; B_0 = S_i - t_{i-1}^*; S_i = \frac{B_0}{2+t_{i-1}^*}$. The amplitude A_0 and the frequency ω the oscillation frequency remains constant. The boundary conditions (7) simulate the conditions of blasting operations in the well, acting at a certain ($x_1 \leq L_1, x_2 = L_3 - l$) distance from the structure D_1 .

Numerical calculations of the problem were carried out with the following initial data: $f(t) = -A_0 t \exp(-t)$ by $t \geq 0; \tau = \Delta t = 0.002; l = 10h; L_1 = 10h; L_2 = 20h; h = \Delta x_1 = \Delta x_2 = 0.05$. External loads $A_0 = -0.5, \omega = 9$. The insertion configuration (region D_1) and the local impact site varied.

Under the described conditions, it is necessary to investigate the stress-strain state of an inhomogeneous medium $D_1 \cap D_2$.

III. METHOD OF BEHAVIORAL CHARACTERISTICS WITH THE IDEAS OF THE SPLITTING METHOD

To solve this problem, along with the initial (1) and boundary conditions (1) - (5), a system of equations consisting of equations of motion and relations of the generalized Hooke's law is used.

$$\rho_k \ddot{u}_\alpha^{(k)} = \sigma_{\alpha\beta}^{(k)} \quad (8)$$

$$\sigma_{\alpha j}^{(k)} = \lambda_k \varepsilon_{\beta\beta}^{(k)} \delta_{\alpha j} + 2\mu_k \varepsilon_{\alpha j}^{(k)}$$

where $\varepsilon_{\alpha j}^{(k)} = 0.5(u_{\alpha, j}^{(k)} + u_{j, \alpha}^{(k)})$; $\delta_{\alpha\alpha}$ – Kronecker symbol, $u_\alpha^{(k)}, \varepsilon_{\alpha j}^{(k)}$ – components of the displacement vector and the strain tensor.

It is convenient to calculate the solution of the problem in a dimensionless space of variables and the desired parameters, which are obtained after the introduction of the notation described in [19].

A. Defining Equations of the Dynamic Problem of the Theory of Elasticity

Using the relations from [19] for dimensionless quantities, one can obtain ($i \neq j$) from equations (8) after simple transformations:

$$\rho_k \dot{v}_\alpha^{(k)} = \sigma_{\alpha\beta}^{(k)}; \dot{\sigma}_{\alpha j}^{(k)} = (\gamma_{11}^{(k)} v_{j, j}^{(k)} + \gamma_{33}^{(k)} v_{i, i}^{(k)}) \delta_{\alpha j} + \gamma_{12}^{(k)} (v_{\alpha, j}^{(k)} + v_{j, \alpha}^{(k)}) (1 - \delta_{\alpha j}) \quad (9)$$

Equations (9) are a linear homogeneous hyperbolic system of first-order differential equations with constant coefficients [16]. Its characteristic surfaces in three-dimensional space ($x_1, x_2; t$) are hypercones with axes parallel to the time axis (Fig. 3).

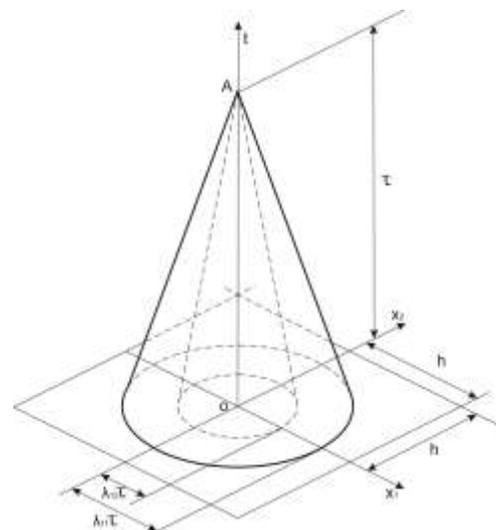


Fig. 3. Characteristic Cones on the Plane.

The system of equations (9) has two families of characteristic cones. These cones coincide with the behavioral characteristics of equations (9). Along the behavioral characteristics lying in the plane $x_i = const$, equations (9) are functions of only two variables ($x_j; t$). This circumstance indicates that the conditions on the behavioral characteristics can be obtained as conditions on the characteristics in the corresponding one-dimensional problem. The corresponding transformations can be performed if one of the spatial variables is alternately fixed in the system of equations (9) [20]. In this case, the system of equations (9) is split into two systems of equations corresponding to the directions $j=1$ and $j=2$ ($i \neq j$):

$$\begin{aligned} \dot{v}_\alpha^{(k)} - \rho_k^{-1} \sigma_{\alpha j}^{(k)} &= A_{\alpha i}^{(k)} \\ \dot{\sigma}_{\alpha j}^{(k)} - \gamma_{\alpha j} v_{\alpha j}^{(k)} &= B_{\alpha i}^{(k)} \end{aligned} \quad (10)$$

where the notation is entered:

$$\begin{aligned} A_{\alpha i}^{(k)} &= \rho_k^{-1} (\sigma_{\alpha\beta, \beta}^{(k)} - \sigma_{\alpha j, j}^{(k)}) \\ B_{\alpha i}^{(k)} &= \gamma_{33}^{(k)} (v_{\alpha, \alpha}^{(k)} - v_{\alpha, j}^{(k)}) \delta_{\alpha j} - \gamma_{12}^{(k)} (v_{i, j}^{(k)} - v_{j, i}^{(k)} - v_{\alpha, j}^{(k)}) (1 - \delta_{\alpha j}) \end{aligned} \quad (11)$$

The differential equations of characteristics have the form:

$$dx_\alpha = \pm \lambda_{\alpha j}^{(k)} dt, \quad (12)$$

and the conditions on the bicharacteristics are:

$$d\sigma_{\alpha j}^{(k)} \pm (-1) \rho_k \lambda_{\alpha j}^{(k)} dv_\alpha^{(k)} = (B_{\alpha i}^{(k)} \pm (-1) \rho_k \lambda_{\alpha j}^{(k)} A_{\alpha i}^{(k)}) dt \quad (13)$$

There $\lambda_{\alpha j}^{(k)} = \gamma_{11}^{(k)}$ if $\alpha=j$ and $\lambda_{\alpha j}^{(k)} = \gamma_{12}^{(k)}$ if $\alpha \neq j$. It can be seen from (12) that on each of the two hyperplanes there are two pairs of families of bicharacteristics that determine the longitudinal $\lambda_{ii}^{(k)}$ and shear $\lambda_{ij}^{(k)}$ ($i \neq j; i, j = 1, 2$) wave propagation velocities (Fig. 3). In each of the two planes ($x_i; t$) there are two families of behavioral characteristics of the positive and negative directions. The upper sign corresponds to the behavioral characteristics of the positive, and the lower sign corresponds to the negative directions. Equations (12) and (13) correspond to each other with the same pair of indices and with the same arrangement of signs. Equations (10) and conditions (13) are used to find a solution to the formulated problem (1) - (5) [23].

B. Selecting the Point Scheme of the Template

To perform numerical calculations of the formulated problem for a region with a given configuration $D_1 \cap D_2$, it is necessary to study the characteristic surfaces. The body $D_1 \cap D_2$ is exposed to non-stationary loads. The initial conditions (1) are given by stresses and displacement velocities in the entire body, and the boundary conditions are given by stresses on the surface (2) - (3). It is assumed that both are continuous differentiable functions. The shape of the body is such that it admits the existence of a coordinate system x_i ($i = 1, 2$), in which the boundary surfaces are coordinate [25].

Let the body $D_1 \cap D_2$ be divided into cells formed by intersections of coordinate surfaces $x_i = const$ ($i = 1, 2$). The

linear dimensions of these cells in the direction of the axes x_1 and x_2 are considered uniform and equal to h . The intersections of the lines $x_i = const$ ($i = 1, 2$) form nodes. At these nodal points, the values of the desired functions $v_\alpha^{(k)}, \sigma_{\alpha j}^{(k)}$ ($\alpha, j = 1, 2$) are found at various time points $t_n - \tau, t_n, t_n + \tau$ ($n = 1, 2, \dots, N$) in time increments τ .

A template consisting of a node O and points $E_{\alpha j}^{\pm(k)}$ lying on the coordinate lines $x_j = const$ and spaced from the point O at distances $\lambda_{11}^{(k)} \tau$ and $\lambda_{12}^{(k)} \tau$ is accepted (Fig. 4). Oblique lines originating from point A are bicharacteristics. In the future, the values of the functions at point O are assigned the upper sign "0"; at points $E_{\alpha j}^{\pm(k)}$ - the lower and upper signs \pm (for example, $\sigma_{\alpha j}^{\pm(k)}$), and at point A an additional index is not assigned [21-24].

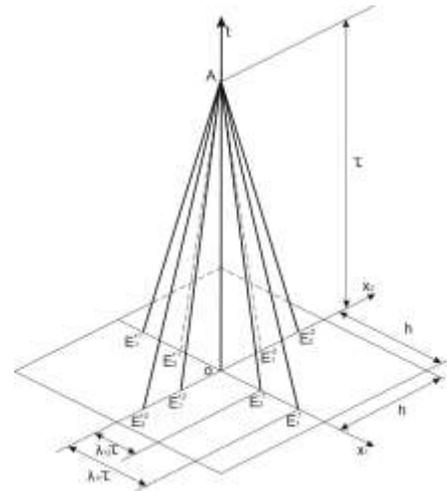


Fig. 4. Point Grid of the difference Scheme for Internal Points.

Based on the described point schemes, the method developed below for solving dynamic problems allows us to determine the particle velocities $v_\alpha^{(k)}$ and the components of the stress tensor $\sigma_{\alpha j}^{(k)}$ at point A on the calculated layer t_n if their values are known on the previous layer t_{n-1} ($n = 1, 2, \dots, N$) at point O and at adjacent points $E_{\alpha j}^{\pm(k)}$. Difference schemes of this type are called explicit. Explicit schemes are convenient because there are no difficulties in solving the systems of difference equations associated with them. These systems are solved sequentially from one-time layer to the next. In this case, the desired values at each node point, unlike the implicit difference scheme, are calculated independently of the others [20].

C. Resolving difference Equations for Solving Dynamic Boundary Value Problems

Resolving difference equations at internal points. The calculation algorithm of the second order of accuracy is constructed below [20]. The integration of the system of equations (10) from point O to point A and the relations (13) from point $E_{\alpha j}^{\pm(k)}$ to point A by the trapezoid method (Fig. 4) allows us to obtain expressions of the following type:

$$\begin{aligned} v_{\alpha}^{(k)} &= v_{\alpha}^{0(k)} + \frac{\tau}{2} (\dot{v}_{\alpha}^{0(k)} + \rho_k^{-1} \sigma_{\alpha j, j}^{(k)} + A_{\alpha i}^{(k)}); \\ \sigma_{\alpha j}^{(k)} &= \sigma_{\alpha j}^{0(k)} + \frac{\tau}{2} (\dot{\sigma}_{\alpha j}^{0(k)} + \gamma_{\alpha j}^{(k)} v_{\alpha j}^{(k)} + B_{\alpha i}^{(k)}) \end{aligned} \quad (14)$$

and

$$\begin{aligned} \sigma_{\alpha j, j}^{(k)} - \sigma_{\alpha j}^{\pm(k)} \pm (-1) \rho_k \lambda_{\alpha j}^{(k)} (v_{\alpha}^{(k)} - v_{\alpha}^{\pm(k)}) &= \\ = \frac{\tau}{2} (B_{\alpha i}^{(k)} + B_{\alpha i}^{\pm(k)} \pm (-1) \rho_k \lambda_{\alpha j}^{(k)} (A_{\alpha i}^{(k)} + A_{\alpha i}^{\pm(k)})) \end{aligned} \quad (15)$$

where the unknown values at point A are taken without additional indexes.

The values of functions at non-node points $E_{\alpha j}^{\pm(k)}$ are replaced by the values calculated by the Taylor formula with first-order accuracy for the functions $A_{\alpha i}^{\pm(k)}$ and $B_{\alpha i}^{(k)}$ and with second-order accuracy for the functions $v_{\alpha}^{(k)}$ and $\sigma_{\alpha j}^{(k)}$ through their values at the node points $O(x_1, x_2; t)$ [20]:

$$\begin{aligned} A_{\alpha i}^{\pm(k)} &= A_{\alpha i}^{0(k)} \pm (-1) \lambda_{\alpha j}^{(k)} \tau \frac{\partial A_{\alpha i}^{(k)}}{\partial x_j}; \\ B_{\alpha i}^{\pm(k)} &= B_{\alpha i}^{0(k)} \pm (-1) \lambda_{\alpha j}^{(k)} \tau \frac{\partial B_{\alpha i}^{(k)}}{\partial x_j} \end{aligned} \quad (16)$$

and

$$\begin{aligned} \sigma_{\alpha j}^{\pm(k)} &= \sigma_{\alpha j}^{0(k)} \pm (-1) \lambda_{\alpha j}^{(k)} \tau \frac{\partial \sigma_{\alpha j}^{0(k)}}{\partial x_j} + \frac{1}{2} (\lambda_{\alpha j}^{(k)} \tau)^2 \frac{\partial^2 \sigma_{\alpha j}^{0(k)}}{\partial x_j^2}; \\ v_{\alpha}^{\pm(k)} &= v_{\alpha}^{0(k)} \pm (-1) \lambda_{\alpha j}^{(k)} \tau \frac{\partial v_{\alpha}^{0(k)}}{\partial x_j} + \frac{1}{2} (\lambda_{\alpha j}^{(k)} \tau)^2 \frac{\partial^2 v_{\alpha}^{0(k)}}{\partial x_j^2} \end{aligned} \quad (17)$$

Partial derivatives of the system of equations (10) with respect to the variable x_j are written as:

$$\begin{aligned} \frac{\partial \dot{v}_{\alpha}^{0(k)}}{\partial x_j} &= \rho_k^{-1} \frac{\partial \sigma_{\alpha j, j}^{0(k)}}{\partial x_j} + \frac{\partial A_{\alpha i}^{0(k)}}{\partial x_j} \\ \frac{\partial \dot{\sigma}_{\alpha j}^{0(k)}}{\partial x_j} &= \gamma_{\alpha j}^{(k)} \frac{\partial v_{\alpha j}^{(k)}}{\partial x_j} + \frac{\partial B_{\alpha i}^{(k)}}{\partial x_j} \end{aligned} \quad (18)$$

Substituting the relations (16), (17) in (15), then excluding with the help of (14) the variables $v_{\alpha}^{(k)}$, $\sigma_{\alpha j}^{(k)}$ and taking into account (18), we can obtain eight equations with respect to the derivatives $v_{\alpha j}^{(k)}$, $\sigma_{\alpha j, j}^{(k)}$, $A_{\alpha i}^{(k)}$, $B_{\alpha i}^{(k)}$:

$$\begin{aligned} \gamma_{\alpha j}^{(k)2} v_{\alpha j}^{(k)} \pm (-1) \lambda_{\alpha j}^{(k)} \sigma_{\alpha j, j}^{(k)} &= \\ = \gamma_{\alpha j}^{(k)2} \left(v_{\alpha j}^{0(k)} + \tau \frac{\partial v_{\alpha}^{0(k)}}{\partial x_j} \right) \pm \lambda_{\alpha j}^{(k)} \left(\sigma_{\alpha j, j}^{0(k)} + \tau \frac{\partial \dot{\sigma}_{\alpha j}^{0(k)}}{\partial x_j} \right) \end{aligned} \quad (19)$$

Adding and subtracting the corresponding pairs of equations (19) in turn, we can find unknown derivatives:

$$\begin{aligned} v_{\alpha j}^{(k)} &= v_{\alpha j}^{0(k)} + \tau \frac{\partial \dot{v}_{\alpha}^{0(k)}}{\partial x_j}; \\ \sigma_{\alpha j, j}^{(k)} &= \sigma_{\alpha j, j}^{0(k)} + \tau \frac{\partial \dot{\sigma}_{\alpha j}^{0(k)}}{\partial x_j} \end{aligned} \quad (20)$$

The system of equations (20) can be used to determine unknown derivatives, both in the internal and boundary nodal points of the studied region $D_1 \cap D_2$. Such expressions can be obtained directly by integrating the system of equations (9) according to the Euler scheme, having previously differentiated them by x_j . However, it is important to have intermediate relations (19), which are used in solving systems of equations where boundary functions are given. The substitution of equalities (20) in (14) allows us to obtain unknown functions $v_{\alpha}^{(k)}$, $\sigma_{\alpha j}^{(k)}$ at the internal nodal points of an inhomogeneous body at time $t_{n-1} + \tau$ ($n = 1, 2, \dots, N$) [21-22].

IV. ANALYSIS OF THE RESULT

Based on the developed information system, the calculation results were obtained, shown in Fig. 5 to 7, when $l = 5h, L_1 = 5h, L_2 = 10h, N_1 = 8h, N_2 = 11h$, calculation results.

Fig. 5 shows the changes in the normal stresses along σ_{22}^k ($k = 1, 2$) the $\frac{x_2}{h}$ coordinate at time $t=60\tau$ in different sections:

1 – $x_1 = 5h$ (this section passes through both the insert and the quarter plane);

2 – $x_2 = 10h$ (the cross-section runs along a quarter of the plane);

2' – $x_1 = 10h$ (the cross-section passes through the insert);

3 – $x_1 = 0h$ (the cross-section passes through the insert).

The normal stresses σ_{22}^k ($k = 1, 2$) in Section 1 near the angular point $N(x_1 = L_1, x_2 = L_3)$, at the boundary of the free surface $PN(x_1 = L_1, l \leq x_2 \leq L_3)$ and in the insertion region ($x_1 = L_1, |x_2| \leq l$) change abruptly with different local extrema. The normal stresses σ_{22}^k in the section 2 in the quarter plane reach the maximum extreme value (-0.22). It is caused by a given load. On the contact surface of the insert and the quarter plane, the normal stresses σ_{22}^k ($k = 1, 2$) suffer a break, taking different values at the insertion points and at the points of the quarter plane. At the exit from the contact area, the normal voltages σ_{22}^k again take on a local extreme value. It should be noted that in the contact region, the value of the extremum of the normal voltage σ_{22}^k (in the Section 2') is almost twice as high as its value in the material of the quarter plane. This result is associated with a greater rigidity of the insert material.

The evolutions of the wave pattern of tangential stresses σ_{12}^k ($k = 1, 2$) at times $t=40\tau$ and $t=60\tau$ are shown in Fig. 6 and 7.

The isolines constructed in the figures are in good agreement with the nature of the change in pulse loads specified at the boundary with a quarter of the plane. At time $t=40\tau$ (Fig. 6), the leading edge of the boundary wave reaches the middle of the insert ($x_1=0$ axis), i.e., the wave has passed the distance $x_2=20h$ with some delay due to the weakening of elastic perturbations during the transition to the insert material. This moment of time $t=40\tau$ corresponds to the action

time of the first pulse of the boundary load. The isolated traces of extensive single extremums in the entire region $D_1 \cap D_2$ of an inhomogeneous medium are explained by the influence of

angular points, points of discontinuity of boundary conditions and contact surfaces.

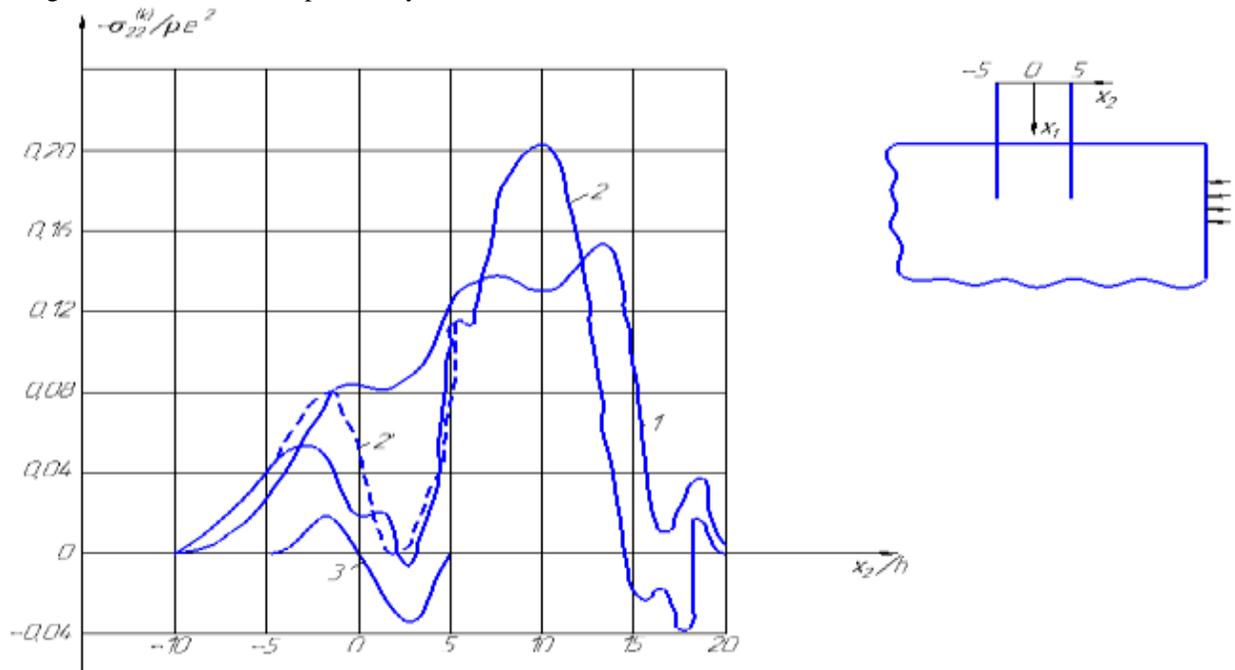


Fig. 5. Change of Normal Stresses σ_{22}^k at Time $t=60\tau$ along the x_2 Coordinate in Sections 1 - $x_1 = 5h$, 2 - $x_2 = 10h$, 2' - $x_1 = 10h$, 3 - $x_1 = 0h$.

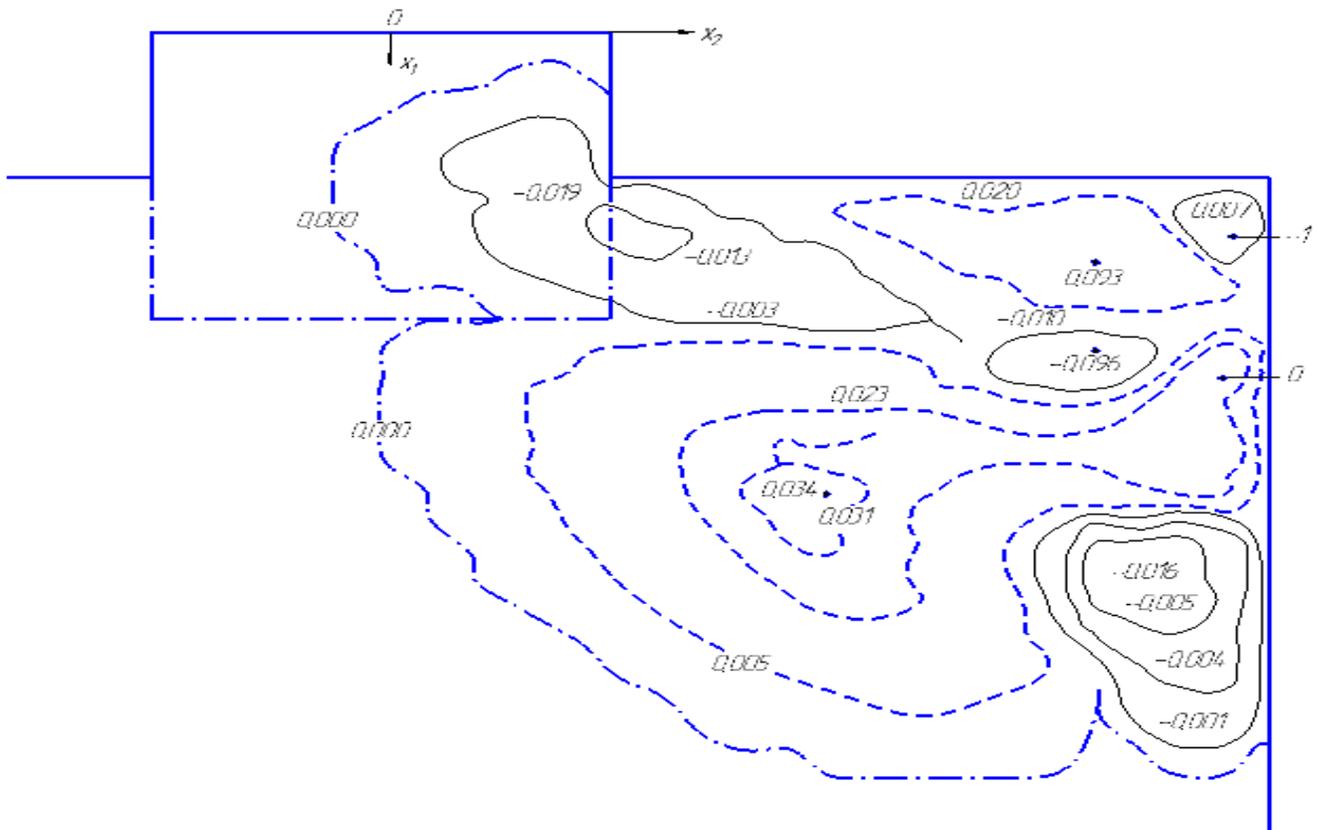


Fig. 6. Tangential Stress Isolines σ_{12}^k ($k = 1, 2$) at Time $t=40\tau$.

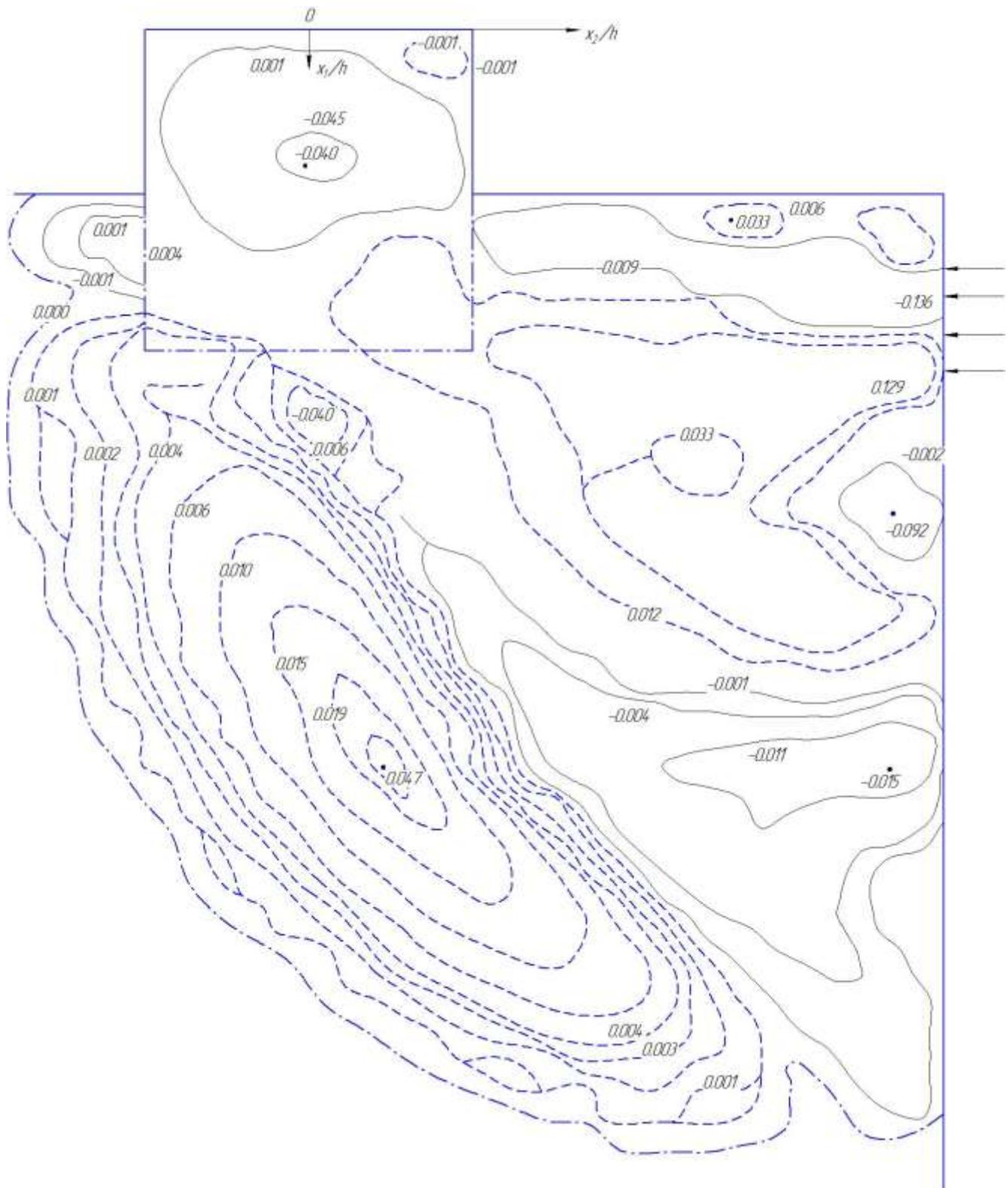


Fig. 7. Tangential Stress Isolines σ_{12}^k ($k = 1,2$) at Time $t=60\tau$.

In Fig. 7, the tangential stresses σ_{12}^k ($k = 1,2$) reach their maximum values in absolute magnitude near the load application points ($N_1 \leq x_1 \leq N_2, x_2 = L_3$; line B_1B_2). At this moment of time $t=60\tau$, the second pulse is affected. This

can explain the appearance of the selected double zones of stress extremes in the entire region of the inhomogeneous medium $D_1 \cap D_2$.

V. CONCLUSION

The developed information system made it possible to visualize the results of calculating wave propagation during blasting operations, to obtain graphs of changes in normal stresses and tangential stress isoclines [26].

Scientific and theoretical interest in the study of the relationship between stress (stress tensor) and deformations occurring in the medium as a result of blasting remains very relevant. The problems of seismically unstable areas require consideration of this issue from the point of view of modeling the wave process.

Modeling the reaction of the soil to changes resulting from an explosion, using information and communication technologies, allows solving problems not only of an economic, construction, geological nature, but also of the safety of human life.

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